

# Airworthiness Directive Schedule

## Aeroplanes

### Beechcraft D18S (Expeditor)

31 August 2000

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- Notes**
1. This AD schedule is applicable to Beechcraft D18S aircraft, manufactured under Federal Aviation Administration (FAA) Type Certificate No. A-765.
  2. The Federal Aviation Administration (FAA) is the National Airworthiness Authority (NAA) responsible for the issue of State of Design Airworthiness Directives (ADs) for these aircraft.  
  
State of Design ADs can be obtained directly from the FAA website at: [Dynamic Regulatory System \(faa.gov\)](http://www.faa.gov)
  3. The date above indicates the amendment date of this schedule.
  4. New or amended ADs are shown with an asterisk \*

### Contents

DCA/B18/1	Airworthiness Directive Compliance at Initial Airworthiness Certificate Issue .....	2
* DCA/B18/2	Cancelled .....	2

**DCA/B18/1      Airworthiness Directive Compliance at Initial Airworthiness Certificate Issue****Applicability:** All Model D18S**Requirement:** Compliance with the following FAA Airworthiness Directives (as applicable) is required:-

- 47-33-05 - Horizontal stabilizer spar
- 47-33-06 - Stabilizer spar replacement
- 47-33-07 - Static selector valve
- 47-34-01 - Expansion type filler cap
- 47-34-02 - Tail cone drain holes
- 47-34-03 - Fuel cell seal
- 48-13-01 - Generator control box
- 48-34-01 - Stabilizer attachment fittings
- 49-29-02 - Rudder spring bushings
- 59-18-02 - Fuel line modification
- 65-03-01 - Gap seal strips
- 66-30-01 - Static pressure line
- 68-23-06 - Volpar tri-gear
- 72-16-02 R3 - Volpar nose gear fork
- 72-25-06 - Nose gear extension
- 75-27-09 R2 - Wing structure
- 77-19-07 - Control column assembly
- 78-10-01 - Lower rudder torque tube

*Note: Each part of this AD (each individual FAA AD) shall be certified in the aircraft log book separately.*

**Compliance:** Before issue of New Zealand Airworthiness Certificate. Repetitive inspections to be accomplished at intervals not exceeding the times specified in the FAA Airworthiness Directives**Effective Date:** 7 May 1999**\* DCA/B18/2      Cancelled**